The application of themodynamic principle to engineering systems Requires data for specific internal energy, enthalpy, entropy and other propostices The thermodynamic relations allow u, h, 5 and other thermodynamic propeeties of simple compressible systems to be evaluated from data that are more readily measured.

Properties used for defining single phase substance

8 properties: P, u, T, s, u, h, g, f. measurable properties are pressure (P), volume (V)

The non measurable properties can be expressed interms of measurable properties.

Thumodynamic Potential - ability to do work.

1. Internal Energy, u

2. Enthalpy, h.

3 Helmholtz potential, f.

4. Cribbs potential, 9

These are the potential (peoperties) of the 8ystem which cannot be measured directly

For a fuce substance in single phase, a theimodynamic potential can be expressed as a function of 2 other properties

### \* Theemodynamic Potentials

#Internal Energy: Suppose a system undergoes a change of state during an adiabatic process. Applying 1st law of Thermodynamics, As a decrease in u' represents the potential to do work, u' is called the modynamic potential

As per 1st law of TD, we know du = dq - dw - 0As per 2nd law of TD,  $\frac{dq}{T} = ds \Rightarrow dq = Tds - 0$ Hence by combining eqns ( + 2) for a Reversible process, we can have. du = Tds - Pdv -> Gibb's First equation

\* Enthalpy

Let us consider a steady flow device (fuebine) Applying 1st law of TD for control volume of turbine, we have, df + me(be + Ve2 + 9 Ze) - mi(hi+V12 +9 Zi) = (0-Wshaft considering Steady flow, KE & Potential energy terms are negligible, device is insulated. we have, Wshaft = - (he-hi)

(2)

We know, h= u+pu. dh = du + Pdo + vdp dh = Tds-Pdv+Pdv+Vdp dh = Tds+vdp -> Gibbs 2nd equation

#### # Helmboltz Potential

Let us consider a system interacting. Surrounding.

As per Ost law of TD, work done by the system during a given process.

dw = dq - du

Let the Burrounding be maintained at constant

temperature To. By and law of TD, dsystem + ds surrounding >0

dS-d2 ≥ 0 -> d2 ≤ To d3 -2

Workdone by the system is given by

dw= d2 - du

dw = Tods - du

W12 = (41-42) - To (S1-S2)

W12 = (U1-TISI) - (U2-T2SZ)

W12 = f1-f2

fi= u1-Tisi -> Helmholtz function potential fz = Uz-TzSz -> Helmholtz function / potential

fi-f2 -> G Helmholtz free energy.

- Idelmholte potential / function f= u-Ts df = du - Tds - SdTBut du = Td3 - PdV df = Tds - Pdv -Tds-SdT df = -sdT - Pdv. -> Itelmholtz free energy equation Let us consider a system which is operating under steady state
flow through a control volume. A Gribb's Free Energy dE + me (het be2+gze) - mi (het bi2+gzi) = 0'- Wabagi. Wish = 9 - (he-hi) -0 If the CV recieves heat energy from surrounding Then  $\Delta S_{surrounding} = \frac{-9}{T_0}$ By and law of 7D, Dssys + Dssur >0 AS - = ≥0 2 ≤ To (Se-Si) - 2 Combining O 40. Wshaft = To (Se-Si) - (Re-Ri) If the process takes place is othermally, Ti=Te=To

Wh 
$$\leq$$
 (hi-Tisi) - (he-Tese)

When  $\leq$  9i - 9e.

 $g = h - Ts \Rightarrow Gibbs function | Polential$ 
 $gi-ge \Rightarrow Gibbs free energy$ 
 $dg = dh - Tds - sdT - 3$ 
 $dh = Tds + vdp - Tds - sdT$ 
 $dg = vdp - sdT - Gibbs free energy ein$ 

Thermodynamic Botenhau	Differential Expression (Specific property)	(Total Property)	Natural Variables
и	du=Tds-Pdv	dU = TdS - PdV	S, Le
h	dh = Tds + vdp	dH=TdS+VdP	s, p
f	df = - Pdu -sdT	dF=-PdV-SdT	V, T
9	dg = Vdp-sdT	dG=VdP-SaT	P, T

$$U = \phi(s, V) \qquad F = \phi(V, T)$$

$$H = \phi(S|P) \qquad G = \phi(P|T)$$

The equations that relate the partial derivatives of properties P, wand, T and 5 of a simple compressible system to each other are called Max well relations. They are obtained by the four cribbs equations by exploiting the exactness of the differentials of thumodynamic peoperties.

Note: If Z = Z(x,y) -> continuous function of ney.

dz = (2z ) dx + (2z ) dy [sm dz = Mdu + Ndy -0 2y,

ie (2M) = (2N)

= 222

we have Gibb's first egn, du=Tds-PdU -@

By analogy, of egns 0 400

we have  $\left(\frac{2T}{2V}\right)_S = -\left(\frac{2P}{2S}\right)_V - \left(\frac{T}{2S}\right)_V$ 

we have , libbs second egn, dH = Tds+Vdp

 $\left(\frac{\partial \Gamma}{\partial p}\right)_{S} = -\left(\frac{\partial P}{\partial S}\right)_{D} - \left(\frac{\partial \Gamma}{\partial S}\right)_{D}$ 

We have Gibbs 3rd eqn, 
$$dF = -PdV - 5dT$$

$$\left(\frac{\partial P}{\partial T}\right)_{V} = \left(\frac{\partial 5}{\partial V}\right)_{T} - \left(\frac{\Pi}{U}\right)$$
We have Gibbs 4th eqn,  $dG_{1} = VdP - 5dT$ 

$$\left(\frac{\partial V}{\partial T}\right)_{P} = \left(\frac{\partial 5}{\partial P}\right)_{T} - \frac{\Pi}{U}$$

Thermodynamic Potential	Natural Variables	Differential Expression	Maxwell Relations
U	5,V		(2T) = -(2P)
Н	S,P	dH=TdS+VdP	(DI) = (DV)
F	V, T		$\left(\frac{\partial P}{\partial T}\right)_{V} = \left(\frac{\partial S}{\partial V}\right)_{T}$
G	PIT		$\left(\frac{\partial V}{\partial T}\right)_{P} = \left(\frac{\partial S}{\partial P}\right)_{T}$

The above 4 selations, are kalled Manuell Relations.

These Relations are extremely valuable in the modynamics because they provide means of the changes in properties (non measurable) determining the changes in properties by simply measuring the phanges in properties by simply measuring the phanges in properties

· Manuvell Relations are limited to simple compressible systems

Note

9f 
$$f(x,y,z) = 0$$
 or  $z = z(x,y)$ 
 $dz = \left(\frac{\partial z}{\partial z}\right) dx + \left(\frac{\partial z}{\partial y}\right) dy$ 

Rules for Partial Derivative Method

1. Reciprocal Relation  $\Rightarrow \left(\frac{\partial x}{\partial y}\right) z = \left(\frac{\partial y}{\partial x}\right) z$ 

2. Cyclic Relation  $\Rightarrow \left(\frac{\partial z}{\partial y}\right) z = \left(\frac{\partial z}{\partial x}\right) y = -1$ 

3. Chain Rule  $\Rightarrow \left(\frac{\partial z}{\partial x}\right) y = \left(\frac{\partial z}{\partial y}\right) y = \left(\frac{\partial z}{\partial y}\right) y$ 

if  $z = z(p,y)$ ,  $p = p(x)$  and  $y = y$ 

4  $v = \left(\frac{\partial v}{\partial z}\right) ds + \left(\frac{\partial v}{\partial y}\right) dv = 0$ 

we have

 $dv = Tds - pdv = 2$ 

Companying  $z = ens$ 

$$dV = \begin{pmatrix} \frac{\partial U}{\partial s} \end{pmatrix}_{V} ds + \begin{pmatrix} \frac{\partial U}{\partial V} \end{pmatrix}_{S} dV \qquad -D$$
we have,
$$dU = TdS - PdV \qquad -2$$

$$companning \quad 2 \quad esns, \quad T = \begin{pmatrix} \frac{\partial U}{\partial s} \end{pmatrix}_{V}; \quad P = \begin{pmatrix} \frac{\partial U}{\partial V} \end{pmatrix}_{S}$$

$$H = \phi \left( \frac{S}{I} P \right)$$

$$dH = \phi \left( \frac{S}{I} P \right)$$

$$dH = \begin{pmatrix} \frac{\partial H}{\partial S} \end{pmatrix}_{P} ds + \begin{pmatrix} \frac{\partial H}{\partial P} \end{pmatrix}_{S} dP$$
we have, 
$$dH = TdS + VdP$$

$$T = \begin{pmatrix} \frac{\partial H}{\partial S} \end{pmatrix}_{P}; \quad V = \begin{pmatrix} \frac{\partial H}{\partial P} \end{pmatrix}_{S}.$$

$$F = \Phi(V_{1}T)$$

$$dF = \left(\frac{\partial F}{\partial V}\right)_{T} dV + \left(\frac{\partial F}{\partial T}\right)_{V} dT$$

$$dF = -P dV - S dT$$

$$P = -\left(\frac{\partial F}{\partial V}\right)_{T}; S = -\left(\frac{\partial F}{\partial T}\right)_{V}$$

$$G_{1} = \Phi(P_{1}T)$$

$$dG_{2} = \left(\frac{\partial G}{\partial P}\right)_{T} dP + \left(\frac{\partial G}{\partial T}\right)_{D} dT$$

$$We have  $dG_{1} = V dP - S dT$ 

$$V = \left(\frac{\partial G}{\partial P}\right)_{T}; S = -\left(\frac{\partial G}{\partial T}\right)_{P}$$

$$T - dS \quad Equations$$$$

& First T-ds Equation S = S(T, V) for single component single phase substance. ds = (DS ) dT + (DS ) dT | du = TdS - PAN Tas = T(25), dT+T(25) dT | dR = dU = Tas. Tas= CraT+Tap dv First Tels Equation

$$C_{V} = \begin{pmatrix} dQ \\ dT \end{pmatrix} V = 0$$

$$dU = TdS - PdV$$

$$dQ = dU + PdV$$

$$dQ = dU = TdS.$$

$$C_{V} = \frac{dQ}{dT} = T\begin{pmatrix} dS \\ dT \end{pmatrix} V$$

$$Also from Manwell Relation$$

$$\begin{pmatrix} \frac{QS}{QV} \end{pmatrix}_{T} = \begin{pmatrix} \frac{QP}{QT} \end{pmatrix}_{V}$$

The second T-ds Equation
$$S = S(T_1P)$$

$$dS = \begin{pmatrix} 25 \\ 2T \end{pmatrix} dT + \begin{pmatrix} 25 \\ 2P \end{pmatrix} dP$$

$$TdS = T\begin{pmatrix} 25 \\ 2T \end{pmatrix} dT + T\begin{pmatrix} 25 \\ 2P \end{pmatrix} dP$$

$$TdS = CpdT - T\begin{pmatrix} 2V \\ 2T \end{pmatrix} dP$$

$$2^{nd} T-ds = Cquation$$

$$\hat{G}_{p} = \begin{pmatrix} dR \\ dT \end{pmatrix}_{p}$$

$$dQ = dU + PdV$$

$$dQ = d(U + PV)$$

$$dQ = dH$$

$$dH = TdS + VdP$$

$$dH = TdS$$

$$dH = Td$$

Third T-ds Equation

$$S = S(P)V)$$

$$dS = \left(\frac{\partial S}{\partial P}\right)_{V} dP + \left(\frac{\partial S}{\partial V}\right)_{P} dV$$

$$dS = \left(\frac{\partial S}{\partial P}\right)_{V} dP + \left(\frac{\partial S}{\partial V}\right)_{P} dV$$

$$TdS = T\left(\frac{\partial S}{\partial T}\right)_{V} \left(\frac{\partial T}{\partial P}\right)_{V} dP + T\left(\frac{\partial S}{\partial T}\right)_{P} \left(\frac{\partial T}{\partial V}\right)_{P} dV$$

$$TdS = C_{V} \left(\frac{\partial T}{\partial P}\right)_{V} dP + C_{P} \left(\frac{\partial T}{\partial V}\right)_{P} dV - T$$

$$\frac{Note}{2}$$

$$Volume Expansivity, \beta = \frac{1}{V} \left(\frac{\partial V}{\partial P}\right)_{T}$$

$$4 \text{sentropic compressibility, } k_{T} = \frac{1}{V} \left(\frac{\partial V}{\partial P}\right)_{T}$$

We have, 
$$B = \frac{1}{V} \begin{pmatrix} \frac{2V}{2T} \\ \frac{2V}{2T} \end{pmatrix}_{p} \rightarrow \begin{pmatrix} \frac{2T}{2V} \\ \frac{2V}{2P} \end{pmatrix}_{T} = \frac{1}{BV} - 2$$

Also we have,  $k_{T} = -\frac{1}{V} \begin{pmatrix} \frac{2V}{2P} \\ \frac{2V}{2P} \end{pmatrix}_{T} = \frac{1}{2V} \begin{pmatrix} \frac{2V}{2P} \\ \frac{2V}{2P} \end{pmatrix}_{T} = 0$ 

From yelli relation,  $g_{f} = f(P, V, T) = 0$ 

$$\begin{pmatrix} \frac{2P}{2V} \\ \frac{2V}{2P} \end{pmatrix}_{T} \begin{pmatrix} \frac{2V}{2P} \\ \frac{2V}{2P} \end{pmatrix}_{V} = -1$$

$$\begin{pmatrix} \frac{2T}{2P} \\ \frac{2V}{2P} \end{pmatrix}_{T} \begin{pmatrix} \frac{2T}{2V} \\ \frac{2V}{2P} \end{pmatrix}_{T} = -4$$

The contraction of the second of the second

#### Difference in Specific Head Capacities

we have, 
$$TdS = C_p dT - T \left(\frac{\partial v}{\partial T}\right)_p dP = 2^{nd} T dS eqn \rightarrow 0$$

$$TdS = CvdT + T \left(\frac{\partial P}{\partial T}\right)_{\mathcal{Q}} dv = \int_{TdS}^{S_T} ds eqn \rightarrow 0$$

$$0 - \Theta \Rightarrow \left(\frac{\partial P}{\partial T}\right)_{\mathcal{Q}} dT - T \left(\frac{\partial V}{\partial T}\right)_{\mathcal{Q}} dP - T \left(\frac{\partial P}{\partial T}\right)_{\mathcal{Q}} dv = 0$$

$$\left(\frac{\partial P}{\partial T}\right)_{\mathcal{Q}} dT = T \left(\frac{\partial V}{\partial T}\right)_{\mathcal{Q}} dP + T \left(\frac{\partial P}{\partial T}\right)_{\mathcal{Q}} dv$$

$$dT = \frac{T}{C_{\mathcal{Q}} - C_{\mathcal{Q}}} \frac{\partial V}{\partial T} + \frac{T}{C_{\mathcal{Q}} - C_{\mathcal{Q}}} \frac{\partial P}{\partial T} dv - 3$$

$$dT = \left(\frac{\partial T}{\partial P}\right)_{\mathcal{Q}} dP + \left(\frac{\partial T}{\partial V}\right)_{\mathcal{Q}} dv - 4$$

$$By analogy$$

The above selation (eqn 5) can be enpressed interms of two other thermodynamic properties conterms of two other thermodynamic properties called volume expansivity (B) and the isothermal compressibility (kg).

we have, 
$$B = \frac{1}{V} \left( \frac{\partial V}{\partial T} \right)_{p}$$
 $k_{T} = -\frac{1}{U} \left( \frac{\partial V}{\partial p} \right)_{T}$ 
 $C_{p} - C_{V} = -T \left( \frac{\partial V}{\partial T} \right)_{p}^{2} \cdot \left( \frac{\partial p}{\partial V} \right)_{T}$ 
 $C_{p} - C_{V} = T_{V} B^{2}$ 
 $k_{T}$ 
 $C_{p} - C_{V} = T_{V} B^{2}$ 
 $k_{T}$ 

Mayer Relation

For an ideal gas, we know  $C_{p} - C_{V} = R$ 

where  $C_{p} \neq C_{V}$  are specific heat capacities

 $P_{200}f$ :

 $C_{p} - C_{V} = -T \frac{g_{V}}{2T} - \frac{\partial p}{\partial V}_{T}$ 
 $C_{p} - C_{V} = -T \frac{g_{V}}{T^{2}} \cdot -\frac{RT}{U^{2}}$ 
 $C_{p} - C_{V} = R$ 
 $C_{p} - C_{V} =$ 

as the absolute temperature approaches zero.

The two specific heats are identical for truly incompressible substance since we = constant The difference between two specific heats is very small and is usually disregarded for substances that are nearly incompressible such as liquids and solids.

A Ratio of specific hear capacity

Assuming Isentopic process, ds = 6

$$C_p = T \left( \frac{2\nu}{2T} \right)_p \left( \frac{2p}{2T} \right)_s - 3$$

We have cyclic relation f(p, v, T) = 0  $\left(\frac{\partial P}{\partial v}\right)_{T} \cdot \left(\frac{\partial v}{\partial T}\right)_{p} \cdot \left(\frac{\partial T}{\partial p}\right)_{V} = -1$ 

8

Using cyclic celations of suitable Rearrangements  $\frac{Cp}{Cv} = \frac{k_T}{K_5} = \Upsilon$ 

\$ Internal Energy Change

For a system undergoes reversible process
between two equilibrium states, change of intunal
enugy, du=Tds-PdV

Tols = CV dT + T (2P) dV

ie du = CvdT + (T(2P) - P) du

we have  $du = \left(\frac{\partial u}{\partial T}\right)_{t} dT + \left(\frac{\partial u}{\partial v}\right)_{T} dv - if u = u(T,v)$ 

From the above esn we have.

 $c_v = \left(\frac{\partial u}{\partial T}\right)_v = \frac{du}{dT}$ 

Also, (Du) = T (DP) - P - Energy egn.

Proof: Internal energy is a twoction of temperature only for ideal gas.

we have du = CvdT + (T &P) v - P) du

For ideal gas, Pu=RT -> pa=RT

" (DP) u= P - P

:. 
$$T(\frac{3p}{2T})_{V} - P = 0$$

:.  $du = Cv dT$ 

?Internal Energy is a function of temperature only.

\*\*N - Change in Enthalpy—

 $dh = Tols + v dp$ 

From and  $Tols = equations$ .

 $Tols = CpdT - T(\frac{3v}{2T})_{p}dp$ 
 $dh = CpdT + [v - T(\frac{3v}{2T})_{p}]dp$ 

we have

 $dh = (\frac{3h}{2T})_{p}dT + (\frac{3h}{2p})_{T}dp$  if  $h = h(T_{1}p)$ 

By analogy, we have

 $Cp = (\frac{3h}{2T})_{p}$ ;  $v - T(\frac{3v}{2T})_{p} = (\frac{3h}{3p})_{T}$ 

\*\*Pool - We have  $dh = CpdT + (v - T(\frac{3v}{2T})_{p}) dp$ 

we have  $dh = CpdT + (v - T(\frac{3v}{2T})_{p}) dp$ 

For ideal gas, Pre=RT -> 10 = RT  $\left(\frac{2P}{2T}\right)_{P} = \frac{P}{P} = \frac{P}{T}$ dh= CpdT + [V - T. 4] dp .. dh = Cpd7 Therefore enthalpy is a function of temperature only for an ideal gas 1 Proof: For isentropic process, PV = C for ideal gas We have, du = Tds - Pdv -> Tds = (VdT+Pdv-1) dh = Tds + vdp -> Tds = CpdT - vdp -@ Fix isenteopic process, ds = 0 (vdT +Pdv=0 → Pdv =- CvdT -3) CpdT-vdp=0 -> udp=CpdT-G (4) => dp + Cp dy = 0 Antegrating. In P+ ln v = Ln C. PV Y = c DEntropy change of an ideal gas. we have, Tds = CvdT + Pdu. Pu=RT ds = CyaT+ FdV P = R ds = CydT · Rdv

Integrating .

$$S_{2}-S_{1} = C_{V} \ln \left(\frac{T_{2}}{T_{1}}\right) + R \ln \left(\frac{V_{2}}{V_{1}}\right) - (T)$$

III's  $TaS = C_{P} dT - V dP$ 

$$dS = C_{P} \frac{dT}{T} - \frac{V}{T} dP$$

$$dS = C_{P} \frac{dT}{T} - R \frac{dP}{P}$$

Integrating  $S_{2}-S_{1} = C_{P} \ln \left(\frac{T_{2}}{T_{1}}\right) - R \ln \left(\frac{P_{2}}{P_{1}}\right) - (T)$ 

Since  $R = C_{P} - C_{V}$ 

$$S_{2}-S_{1} = C_{P} \ln \left(\frac{T_{2}}{T_{1}}\right) - \left(C_{P}-C_{V}\right) \ln \left(\frac{P_{2}}{P_{1}}\right)$$

$$S_{2}-S_{1} = C_{P} \ln \left(\frac{T_{2}}{T_{1}}\right) - \ln \left(\frac{P_{2}}{P_{1}}\right) + C_{V} \ln \left(\frac{P_{2}}{P_{1}}\right)$$

$$S_{2}-S_{1} = C_{P} \ln \left(\frac{V_{2}}{V_{1}}\right) + \left(V \ln \left(\frac{P_{2}}{P_{1}}\right) - \left(\frac{V_{2}}{P_{1}}\right)\right)$$
Any one of the above  $S_{1}$  equations Lamber when  $S_{2}$  equations Lamber  $S_{2}$  used for computing the entropy change

be used for computing the enterpy change between any two states of an ideal gas.

Entropy change is Polyteopic peocess

9n a seversible adiabatic (isentropic) process entropy elemains bonstant. But in a seversible polytropic process, the entropy changes we have,  $S_2 - S_1 = Cv \ln\left(\frac{T_2}{T_1}\right) + R \ln\left(\frac{v_2}{v_1}\right)$   $S_2 - S_1 = \frac{R}{Y-1} \ln\left(\frac{T_2}{T_1}\right) + \frac{R}{N-1} \ln\left(\frac{T_1}{T_2}\right)$   $S_2 - S_1 = \frac{N-Y}{(Y-1)(N-1)} R \ln\left(\frac{T_2}{T_1}\right) - 1$   $111^{ly} S_2 - S_1 = \frac{N-Y}{N(Y-1)} R \ln\left(\frac{p_2}{p_1}\right) - 2$   $S_2 - S_1 = -\frac{(N-Y)}{Y-1} R \ln\left(\frac{v_2}{v_1}\right) - 3$ 

A JOULE KELVIN EFFECT

Note: When a fluid passes through a restriction such as poeous plug, a capillary tube on an such as poeous plug, a capillary tube on an oldinary valve, its pressure drops. The enthalpy of the fluid remains approximately constant during the process called throttling.

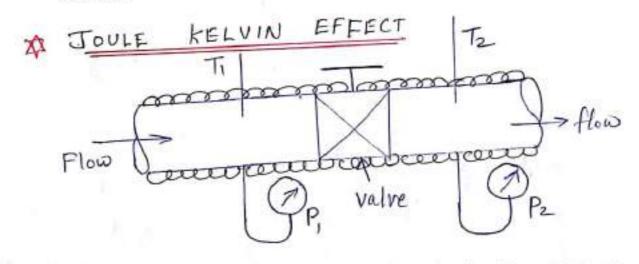
The fluid may enperience large the fluid may enperience large to its temperature as a result of deep in its temperature as a result of the other which form the basis of operation theothers which form the basis of operation of refrigerators and air conditioners.

The temperature may remain same or even

Two counterweighing effects which decide whether temperature increase as decrease during throttling process

1. Temperature will decrease because of its decrease in pressure following equation of state.

of the mechanical energy is converted to intermolecular energy, thereby temperaline invested



A gas is made to undergo continuous—throttling process. The pressure and temperature of the gas in the insulated pipe upstream and downstream of the valve are and measured with suitable manometres and themperature.

Let PI, TI be arabitrarily chosen pressure and temperature before throttling and is kept constant By operating the valve manually, the gas is throttled successively to different pressure and temperature (P2', T2'); (P2", T2") ... These are then plotted on T-P co-ordinates. as shown in figure below. State after All the points represent equilibrium states of some constant mass of gas at which gas has same throttling enthalpy. The were passing through all these points is an isenthalpic cuere or an rsenthalpe The initial and temperature and pressure of the gas are then set to new values and to different states, a family of by throttling isenthalpe are obtained for gas. Trans curve Cooung MITED fi=c Scanned by CamScanner

The temperature behavious of a fluid during a throttling (h=constant) process is described by Joule Thomson coefficient (UJT)

isenthalpe on a T-P diagram at any point is called Joule Thomson on Joule telving coefficient ie 44 is a measure of the change in temperature with pressure during a constant enthalpy prouss.

 $\mu_{JT} = \left(\frac{2T}{2P}\right)_{R}$ 

MJTLO -> Temperature increases.

MJT = 0 -> Temperature remains constant

MJT > 0 -> Temperature decreases.

Some constant enthalpy lines on the T-P diagram pass through a point of zero Joule Thomson coefficient or zero alope. The locus of all these points at which The locus of all these points at which MJT is zero is called Inversion curre of Anversion line.

The region inside the inversions cueve where upt is the inversion outre.

The region outside the inversion overe when upt is -ve is called healing region.

The temperature

Anversion Temperature: The temperature at a point where constant enthalpy line intersects the inversion were time is called Anversion temperature.

\* Expression of Joules Thomson coefficient

MJ = (ST)

we know, dh = Tds + Vdp

= dh = GaT + (V-T (V-T)p) dp

9+ +=h(T, P) dh=(2h)dT+(2h)dP Tas = dh - udp

Tas= Cpat-Tov) dp

For constant enthalpy process (throtting)

$$M_{JT} = \left(\frac{\partial T}{\partial P}\right)_{5} = \frac{1}{C_{p}} \left(T \left(\frac{\partial V}{\partial T}\right)_{p} - V\right)$$

For an ideal gas, MJT = 0

#### Additional Information

Fox achieving the effect of cooling by Joule kelvin expansion, the initial temperature of the gas must be below the point where the inversion over intersects the temperature onus, it below the manimum inversion inversion inversion

Tox nearly all substances, the manimum fox nearly all substances, the normal inversion temperature is above the normal ambrent temperature and kence cooling can be obtained by the Joule kelvin effect. be obtained by the Joule kelvin effect. In the case of 1/2 and life, the gas is In the case of 1/2 and life, the gas is first precoded in a heat exchanges below first precoded in a heat exchanges

CLAPEYRON EQUATION

Note: More well Relations have far reaching implications in their odynamics and are implications in desire useful their odynamic frequently used to desire useful their odynamic selations.

Clapeyron Equation or relations derived from Maxwell relations enables in to from Maxwell relations enables in to determine the enthalpy change associated determine the enthalpy of such as enthalpy of voto weisation, high from a knowledge of vopoweisation, high from a knowledge of

liquid fz,

The Chapeynon relation enables

us to determine enthalpy change associated with a phase change ( such as enthalpy of vapoueisation hgg) from a knowledge of P, v and

T alone

het us lonsides a single component system containing 2 phases (L4V) in a state of equilibrium. For a seversible process at constant T& P. the Gibbs function remains Constant

For phase transition from liquid state to vapour state,

$$g_{f_i} = g_{g_i}$$

In liquid phase,

In gaseon phase

$$\frac{dP}{dT} = \frac{Sg - Sf}{Vg - Vf} = \frac{Sfg}{Vfg}$$

we have 
$$fg = \frac{hfg}{T}$$
, where  $hfg - enthalpy g$ 

$$\frac{(2p)}{2T}_{sat} = \frac{hfg}{T}$$

$$\frac{T}{T}_{sat} = \frac{hfg}{T}_{sat}$$

This Clapeyron regulation enables us to determine the enthalpy of vapourisation high at a given temperature by simply measuring the solution curve on a p-Thingmon and the specific volume of saturated liquid and saturated vapour at the given temperature and saturated vapour at the given temperature phase change process that occurs as sonstant temperature and pressure.

Clapeyron equation can be simplified for liquid - vapour and solid - vapour phase changes by utilizing some approximation.

By utilizing some approximation.

At low pressures,  $v_g >> v_f$  and thus  $v_f = v_g$ .

By lieating the vapour as an ideal gas, we have  $v_g = RT/p$ :

Substituting these approximations into Clapeyron substituting these approximations into Clapeyron equation, we get

$$\left(\frac{dP}{p}\right)_{sat} = \frac{hg}{R} \left(\frac{dT}{T^2}\right)_{sat}$$

For small temperature intervals, Hyg can be treated as constant.

In  $\left(\frac{\beta_2}{p_1}\right)$  sat  $=\frac{h+g}{R}\left[\frac{1}{7_1}-\frac{1}{7_2}\right]$  sat

This equation is called Clapeyron - Clausius equation and it can be used to determine the variation of saturation pressure with temperature.

from 3rd Manwell Relations also.

Real Gas Equation of State Other than Vander Warls equation (Mod 5) 1. Bertheld , P = RT - a - D 2. Dietereci egns of state,  $P = \frac{RT}{11-L} e^{-a/RTv}$ 3 - Redlich - knoong ean of state, P= RT - a T/2. V(v+b) The longtants a and b are evaluated Boom contical data. Berthelot and Dietereci equations of state; like the vanderwaals equation of state has united accuracy. But the Redlich- kwong equation gives good evults at high pressure and is fairly accurate for temperature above critical value

Combustion: A chemical process in which.

fuel is reacted with oridise's while releasing large quantity of energy (Reat & light).

Note: When a chemical reaction occurs, the bonds within the molecules of the reactants are broken and atoms and electrons reassange to form products:

(Methanic eg. CH4+202 > 102+2H20+AH Reactants

(O2+H20)

Imol + 2 mol | 1 mol + 2 mol (Combustion chamber)

Methanic O2 (Combustion chamber)

Note: The three major combustible chemical elements in most common fuels are C, Hz and sulphur

Combustion is complete when all the carbon present in fuel is burned to LOZ, all the hydrogen is burned to burned to burned to burned to burned and all other combustable elements sulphur dionide and all other combustable elements are fully oxidised.

Reactants -> products.

Fuel + oxidiser -> products.

A combustible substance. Hydro carbon fuel | Liquid > gasoline, diesel fuel, kerosene > braseous > CH4 (Methane)

Solid form -> coal. MODELLING COMBUSTION AIR Oxygen is required in every combustion heactions. Puce oxygen is used only in special applications such as cutting and welding. Mostly in combustion applications, are is used as an oudises CH4 + 2 (02 + 79/2 N2) -> (02+2H20+7:52N2 AIR FUEL RATIO 4 FUEL-AIR RATIO These parameters used to quantify the amount of fuel and air in a particular combustion process. , based on mass

Air-fuel ratio = mass of and, based on mass basis

Air-fuel ratio = mass of fuel.

Air-fuel ratio = mole of air Mair

mole of fuel. Mfuel

mole Basis:

(16)

Fuel-Air Ratio = \_\_\_\_\_\_ Air-fuel Ratio.

Stoichiometric Air or Theoretical Air

The minimum amount of air that supplies sufficient oxygen for the complete combustion of all the carbon, Hydrogen and sulphus present in the fuel.

But normally, the amount of air supplied is either quater or less than the theoretical amount The amount of air actually supplied is commonly expressed in teems of the percent of theoretical air.

eg. 150% theoretical and implies the actual and supplied is 15 times the theoretical amount of

The amount of air supplied can be expressed alternatively as a percent excess or

percent deficiency of air:

150% theoretical air > 50% encess air

90% theoretical air > 10% deficiency of air.

#### Equivalence Ratio (Ф)

It is the satio of the actual fuel and sation of saturation of the shickiometric fuel and sation for complete combustion.

φ = (F/A) actual (F/A) stoichiometric'

 $\phi = 1 \rightarrow operating$  at shickeinetric condition  $\phi \geq 1 \rightarrow Fuel$  lean  $\rightarrow less$  fuel.  $\phi \geq 1 \rightarrow Fuel$  lean  $\rightarrow more$  fuel.

### Theoretical lombustion Process

The ideal combustion process the during which a fuel is buent completely with theoretical air is called stoichiometric or theoretical combustion of that fuel.

Theoretical combustion of the fuel.

Theoretical combustion no unbuent fuel

Theoretical combustion no production process

In actual Combustion process it is common practice to use more aix than the stoichiometric amount to incease the chances of complete combustion or to control the temperature of combustion chamber. Note: Orsat Gas Analyser. It is used to

analyse the composition of combustion gases.

Enthalpy of Formation (compound)

It is the amount of energy Released or absorbed when I mole of a Compound is from its respective stable elements the compound and elements all being at Tref and Pref. (standard state).

 $C + O_2 \rightarrow CO_2$ This reaction is escothermu, So for the CO2 to excit 02 | Reactor Tref, Pref at the same temperature as entering elements, Dhf = DHR= hco2-(hc+ho2) there should be heat transfer from reactor to swoundings-Dif = DHR = To compound of formation sty - molar enthalpy Reaction

DHR - Enthalpy of

Note: An enthalpy daltim for the study of macting systems can be established by assigning arbitrary a value of zero to the enthalpy of the stable elements at a state called the standard reference state (Try = 25°C & Proof = 1 atm).

Evaluating Enthalpy: The specific enthalpy of a compound at a state other than the standard state is found by adding the specific enthalpy change Dh between the standard state and state of interest to the enthalpy of formation.

h = hg + sh

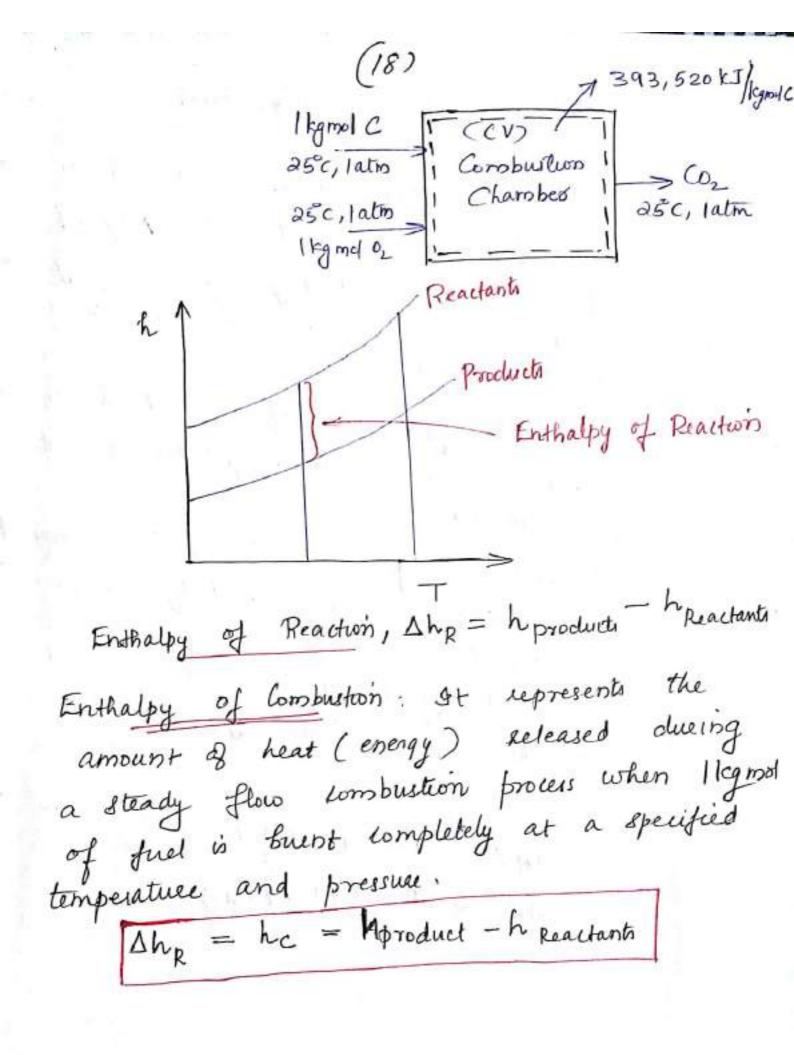
Enthalpy of Reaction (DhR)

9t is defined as the difference
between the enthalpy of product at a

Specified state and enthalpy of reactant

At the same state for a complete reaction

from order gallers.



## Heating Value of the Fuel.

Meating value is defined as the amount of heat released when I kg mol of fuel is buent completely in a steady flow procen while products and reactants are at a specified state.

The heating value of a fuel is a positive number equal to the magnitude of enthalpy of combustion.

Higher Heating value: 4t is obtained when all the water formed by combustion is a liquid.

Lower Heating value: It is obtained when all the water formed by combustion is a vapour.

HHV = LHV + Energy regd to vaponise the liquid

HHV = LHV + (mi htg) Hzo

# Adiabatic Flame Temperature / Adiabatic Combushin Temperature

In the absence of any work interactions and any change in kinetic and potential and any change in kinetic and potential energy, the chemical energy released during a combustion process is either lost as heat a combustion process is either lost as heat to succumdings on is used internally to saise the temperature of combustion products the temperature of combustion products. The smaller the heat loss, larger the temperature list.

Fuel Combustion Product
Reactants (Reactor)

The limperature of a combustion chamber becomes monimum when combustion is complete and no head is lost to the sussundings (Q=0). This manimum temperature achieved in the limit of adiabatic operation of the leador is called adiabatic flame on adiabatic combustion temperature. Adiabatic flame temperature can be determined by setting, W=0,0=0

Q-W= H product - H Reactanto

H products = H Reactanto